# **PCT**

# WORLD INTELLECTUAL PROPERTY ORGANIZATION International Bureau



#### INTERNATIONAL APPLICATION PUBLISHED UNDER THE PATENT COOPERATION TREATY (PCT)

(51) International Patent Classification 6:

B29C 43/36, 33/40, B30B 15/06 // B29K
21:00, 105:10

(11) International Publication Number: WO 98/50214

(43) International Publication Date: 12 November 1998 (12.11.98)

(21) International Application Number: PCT/GB98/01270

(22) International Filing Date: 30 April 1998 (30.04.98)

(30) Priority Data: 9709011.2 3 May 1997 (03.05.97) GB

(71) Applicant (for all designated States except US): ADVANCED COMPOSITES GROUP LTD. [GB/GB]; Composites House, Adams Close, Heanor Gate Industrial Estate, Derbyshire DE75 7SW (GB).

(72) Inventor; and

(75) Inventor/Applicant (for US only): SLOMAN, Roger, Mark [GB/GB]; Advanced Composites Group Ltd., Composites House, Adams Close, Heanor Gate Industrial Estate, Derbyshire DE75 7SW (GB).

(74) Agent: CAMPBELL, Iain, Angus; Swindell & Pearson, 48 Friar Gate, Derby DE1 1GY (GB).

(81) Designated States: AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CU, CZ, DE, DK, EE, ES, FI, GB, GE, GH, GM, GW, HU, ID, IL, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TR, TT, UA, UG, US, UZ, VN, YU, ZW, ARIPO patent (GH, GM, KE, LS, MW, SD, SZ, UG, ZW), Burasian patent (AM, AZ, BY, KG, KZ, MD, RU, TJ, TM), European patent (AT, BE, CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE), OAPI patent (BF, BJ, CF, CG, CI, CM, GA, GN, ML, MR, NE, SN, TD, TG).

**Published** 

With international search report.

(54) Title: IMPROVEMENTS IN OR RELATING TO PRESSURE TRANSMITTERS FOR USE IN THE PRODUCTION OF COMPOSITE COMPONENTS

#### (57) Abstract

A pressure transmitter for use in the curing of composite components may take the form of a caul plate, pressure intensifier or vacuum bag and is formed from a curable elastomeric material which is partially cured, normally when supported on a mould, at a temperature not exceeding 100 °C. It is subsequently further cured at, for example 175 °C, while unsupported. It may incorporate a reinforcing layer formed from a curable resin pre-impregnated fibrous material.

# FOR THE PURPOSES OF INFORMATION ONLY

Codes used to identify States party to the PCT on the front pages of pamphlets publishing international applications under the PCT.

AL	Albania.	ES	Spain	LS	Lesotho	SI	Slovenia
AM	Armenia	FI	Finland	LT	Lithuania	8K	Slovakia
AT	Austria	FR	Prance	LU	Luxembourg	SN	Senegal
AU	Australia	GA	Gabon	LV	Latvia	<b>SZ</b>	Swaziland
AZ	Azerbaijan	GB	United Kingdom	MC	Monaco	TD	Chad
BA	Bosnia and Herzegovina	GE	Georgia	MD	Republic of Moldova	TG	Togo
BB	Barbados	GH	Ghana	MG	Madagascar	TJ	Tajikistan
BB	Belgium	GN	Guinea	MK	The former Yugoslav	TM	Turkmenistan
BF	Burkina Faso	GR	Greece		Republic of Macedonia	TR	Turkey
BG	Bulgaria	HU	Hungary	ML	Mali	TT	Trinidad and Tobego
BJ	Benin	IB.	Ireland	MN	Mongolia	UA	Ukraine
BR	Brazil	IL	Israel	MR	Mauritania	UG	Uganda
BY	Belarus	IS	Iceland	MW	Malawi	US	United States of America
CA	Canada	PT	Italy	MX	Mexico	UZ	Uzbekistan
CF	Central African Republic	JP	Japan	NB	Niger	VN	Viet Nam
CG	Congo	KB	Kenya	NL	Netherlands	YU	Yugoslavia
СН	Switzerland	KG	Kyrgyzstan	NO	Norway	ZW	Zimbabwe
CI	Côte d'Ivoire	KP	Democratic People's	NZ	New Zealand		
CM	Cameroon		Republic of Korea	PL	Poland		
CN	China	KR	Republic of Korea	PT	Portugal		
CU	Cuba	KZ	Kazakstan	RO	Romania		
cz	Czech Republic	I.C	Saint Lucia	RU	Russian Federation		
DE	Germany	u	Liechtenstein	SD	Sodan		
DK	Denmark	LK	Srl Lanka	SE	Sweden		
EE	Estonia	LR	Liberia	SG	Singapore		

# Improvements in or relating to pressure transmitters for use in the production of composite components

This invention relates to pressure transmitters for use in the production of composite components. The term "pressure transmitter" as used herein is intended to refer to cover members and ancillaries used to allow pressure to be applied to composite components during their moulding in a tool. The expression includes caul plates, pressure intensifiers and reusable rubber bags.

The use of caul plates, pressure intensifiers and rubber bags to produce smooth surfaces on composite components is a normal practice. Caul plates and/or pressure intensifiers can be made from an elastomeric material and enable appropriate pressure to be applied to the laminate in order to provide a relatively smooth surface on the bag side. In the case of pressure intensifiers, these are mainly used in female features on the non moulded side of the component. The caul plates consist of an elastomeric sheet which may be reinforced by a resin pre-impregnated fibre. The disadvantage of these caul plates made from fully formulated sheet rubber is that they require to be cured at high temperatures (greater than 100°C). Also, some reinforced caul plates exhibit the disadvantage that the elastomeric material tends to separate from the fibre reinforcement, and others may give undesirable contamination problems.

As used herein, the expression "partially cured" when referring to a curable component means cured to the extent that the component can be removed from a mould, tool or former (which may or may not have high temperature capability (ie being capable of withstanding temperatures of greater than 100°C)), and then be handled and further processed (including a higher temperature further curing stage) without causing damage to the component. Further the expression "elastomeric material" includes the precursors of such materials, for example, fully formulated polymeric materials which after partial or full cure exhibit elast meric properties. The polymeric material may be in sheet form.

Acc rding to a first aspect of the present invention there is provided a pressure transmitter comprising an elastomeric material which is partially r at least partially cured at a temperature not exceeding 100°C.

According to a second aspect of this invention, there is provided a pressure transmitter for use in the production of composite components, the pressure transmitter comprising an inner region of a reinforcement and an outer region of an elastomeric material, the elastomeric material being partially or, at least partially, curable at a temperature not exceeding 100°C.

Preferably, the elastomeric material is partially or, at least partially, curable at 20 - 70°C, preferably 65°C. Conveniently the elastomeric material is partially cured during a period of 14 hours.

Desirably, the elastomeric material is further cured at an elevated temperature to reach a state of being substantially fully cured. Preferably the elevated temperature lies in the range 170 - 185°C. Preferably, the elastomeric material can be further cured at a temperature of substantially 175°C. Conveniently the elastomeric material is further cured for a period of at least 2 hours.

Preferably elastomeric material is a silicone elastomer, an acrylate or a fluoroelastomer.

According to a further aspect of this invention, there is provided a pressure transmitter for use in the production of composite components, the pressure transmitter comprising an inner region of a curable reinforcement and an outer region of elastomeric material, the curable reinforcement comprising resin pre-impregnated onto fibres, said resin being partially, or at least partially, curable at 100°C or less.

Preferably the elastomeric material cures or at least partially cures at a temperature greater than the cure r partial cure temperature of the curable

reinforcement. This provides the advantage that the resin cures or partially cures bef re the elastomer, thereby reducing shrinkage of the elastomer.

Preferably, the elastomeric material is partially, or at least partially, cured at a temperature in the range of 20 - 70°C more preferably at 65°C. Preferably, the elastomeric material is partially cured during a period of 14 hours.

Desirably the elastomeric material is further cured to reach a state of full cure at a temperature in the range of 170 - 185°C. Preferably, the elastomeric material can be further cured at a temperature of substantially 175°C. Conveniently the elastomeric material is further cured during a period of 2 hours.

According to a still further aspect of this invention, there is provided a pressure transmitter for use in the production of composite components, the pressure transmitter comprising an inner region of a curable prepreg reinforcement and an outer region of an elastomeric material, the curable prepreg reinforcement being in the form of resin pre-impregnated fibres defining a plurality of interstitial spaces, wherein the elastomeric material of the outer region substantially fills the interstitial spaces.

This has the advantage that the elastomer is keyed to the fibres, thereby mitigating any separation of the resin pre-impregnated fibres from the elastomer.

Preferably, the pressure transmitter is in the form of a caul plate, or a pressure intensifier for use in applying pressure to surfaces of a moulded component. The invention is particularly suitable for use with moulding techniques which involve the use of a bag applied to the moulding, for example by using vacuum bags.

Acc rding to another aspect of this invention, there is provided a method

of curing a pressure transmitter, comprising providing a pressure transmitter as described above, and thereafter at least partially curing the pressure transmitter at a temperature not exceeding 100°C.

The pressure transmitter may be partially or, at least partially, cured at a temperature in the range of 60 - 70°C (approximately 140 - 160°F), more preferably at 65°C (approximately 150°F). Conveniently, the pressure transmitter is partially cured during a period of at least substantially 14 hours. The pressure transmitter may be partially cured within 14 hours.

Desirably, after said partial or, at least partial, curing, the pressure transmitter is further cured at a temperature in the range of 170 - 185°C (approximately 340 - 365°F). Preferably, the pressure transmitter is further cured at a temperature of substantially 175°C (approximately 350°F). The pressure transmitter may be so further cured during a period of at least substantially 2 hours. The pressure transmitter may be so further cured within 2 hours.

The elastomeric material may be a silicone elastomer.

By appropriate selection of the resin and elastomeric material, the low temperature curing of the pressure transmitter will result in the resin curing or, partially curing, before the elastomer has fully cured. This has the advantage of minimising shrinkage of the elastomer. Another advantage of the low temperature curing (ie less than 100°C) is that it can take place on a tool that can only withstand such low temperatures. The step of further curing the pressure transmitter provides the elastomeric material with the properties to enable the pressure transmitter to perform its function.

An embodiment of the invention will now be described by way of example only with reference to the accompanying drawings in which:-

Fig. 1 shows a schematic diagram of a pressure transmitter in the form

of a caul plate in use;

Fig. 2 shows a schematic diagram of a pressure transmitter in the form of a pressure intensifier in use;

Fig. 3 shows a reinforcement not to scale with resin pre-impregnated fibres defining interstitial spaces; and

Fig. 4 is a view similar to Fig. 3 but showing an elastomer in the interstitial spaces.

Referring to Fig. 1, there is shown a prepreg lay-up 10 to be used in forming a composite component. The prepreg lay-up is arranged on a tool 12 with a caul plate 14 placed on the lay-up 10. The caul plate 14 comprises an inner region 16 of a fibre reinforcement and an outer region 18A,18B of an elastomeric material 18. In the formation of the composite component, a vacuum bag or sheet 19 is arranged over the lay-up 10 and the caul plate 14. Air is removed from the vacuum bag 19 whereby air pressure provides the necessary pressure for moulding and heat is also applied, in the standard way.

The fibre reinforcement 16 comprises woven fibres 20 defining a net like structure having interstitial spaces 22 (see Fig. 3). The interstitial spaces 22 are, as shown in Fig. 4, filled with the elastomeric material 18 defining outer regions 18A,18B. The reinforcement 16 may be formed of one or a plurality of layers of the woven fibres 20.

The caul plate 14 is first processed to make it usable as described in the preceding two paragraphs by partially curing the caul plate 14 at a temperature of less than substantially 65°C at least 14 hours during a first curing stage the caul plate being supported on a mould which need not be capable of withstanding high temperatures (eg above 100°C) during the process. This first curing stage partially r fully cures the resin and partially cures the

elastomeric material 18. This minimises the shrinkage of the elastomeric material 18. After the first curing stage, the caul plate 14 is then further cured during a second curing stage for at least 2 hours at a temperature of substantially 175°C to fully cure the resin and the elastomeric material such that the caul plate 14 develops the appropriate properties normally expected of a caul plate 14 (see below). This second stage does not necessarily require the use of a mould to support the caul plate.

Thus, the resin and elastomeric material are selected to be capable of partial curing at low temperature ie less than 100°C.

As can be seen at Fig. 1 a wrinkle 24 in the prepreg lay-up 10 can be formed where adjacent plies overlap and at other places. The use of a caul plate 14 assists in smoothing out or preventing the formation of such wrinkles 24.

The caul plate may have a microporous membrane 118 arranged on its side adjacent the lay-up 10 to facilitate removal of gases emitted by the resin during the curing process. Alternatively a solid inpermeable sheet for example of HALAR could be substituted for the microporous membrane. The sheet may be in the form of a film bonded to the caul plate.

Referring to Fig. 2, there is shown a tool 12 at a corner region 22 thereof where the prepreg lay-up 10 defines a concave region 24. A pressure intensifier 26 is arranged in the concave region 24 to ensure that appropriate pressure is applied thereto.

The pressure intensifier 26 is formed in a similar manner to the caul plate described above using similar cure and postcure temperatures.

In a modification the reinforcement may be omitted from the caul plate or from the pressure intensifier. In the modification, the caul plate and pressure intensifier are manufactured in the manner described above, that is they are partially cured at a temperature n t exceeding 100°C on a mould which

7

need not be capable of withstanding temperatures f greater than 100°C so that it can be manufactured, for example, from wood, plaster or other suitable easily worked material but after the initial cure stage the caul plate and pressure intensifier is removed from the mould for further curing to a fully or approaching fully cured state at an elevated temperature for example 175°C. As before the initial cure time can be 14 hours and the postcure time 2 hours.

In a further modification of the present invention the pressure transmitter may take the form of a vacuum bag, that is a bag formed, in the simplest example, from two sheets of elastomeric material, sealed or sealable around the periphery of the sheets to form a bag which, after fitting with normal backing connections etc. be used in a vacuum bag moulding process. In this modification, as before, the vacuum bag is formed from two sheets of fully formulated polymeric material which are made into the bag when the material is in an uncured state (examples of suitable material being set out below). The bag can be cured during the initial cure stage which is carried out, for example for 14 hours, at temperatures no greater than 100°C and typically 65°C, thereafter, as appropriate, the bag can be fully cured bearing, for example, a further 2 hour cure cycle at an elevated temperature for example 175°C.

Standard vacuum bag fitments for example vacuum terminals, openable seals etc. can be fitted before the initial cure or if convenient before the postcure, that is after the initial cure.

Examples of the properties of suitable elastomeric materials 18 are set out below.

#### Example 1

#### Physical Properties

Form: soft and conformable with the slightest degree of

tack.

Color: royal blue

8

Thickness: 0.060", (60mils), other thickness will be available.

Width: up to 50", calendered rolls.

## **Cured Properties**

Durometer 70 (nominal)

Tensile Strength 1000+(psi)

Elongation at Break 350+(%)

Modulus at 100% 505 (psi)

Modulus at 300% 940 (psi)

Tear Strength 160+(ppi)

Compression Set 30 (%) (22 hours at 350°F)

Adhesion excellent (cocured to LTM prepreg)

Temperature Resistance 450°F to 500°F (when fully cured)

## Example 2

## **Physical Properties**

Form: soft and conformable with the slightest degree of

tack.

Color: clear

Thickness: 0.085", (85mils), other thickness will be available.

Width: up to 50", calendered rolls.

## **Cured Properties**

Durometer 40 (nominal)

Tensile Strength 1510+(psi)

Elongation at Break 925+(%)

Modulus at 100% 300 (psi)

Modulus at 300% 300 (psi)

Tear Strength 150+(ppi)

9

Compression Set TBD

Temperature Resistance 450°F to 500°F (when fully cured)

Various modifications can be made without departing from the scope of the invention.

## **CLAIMS**

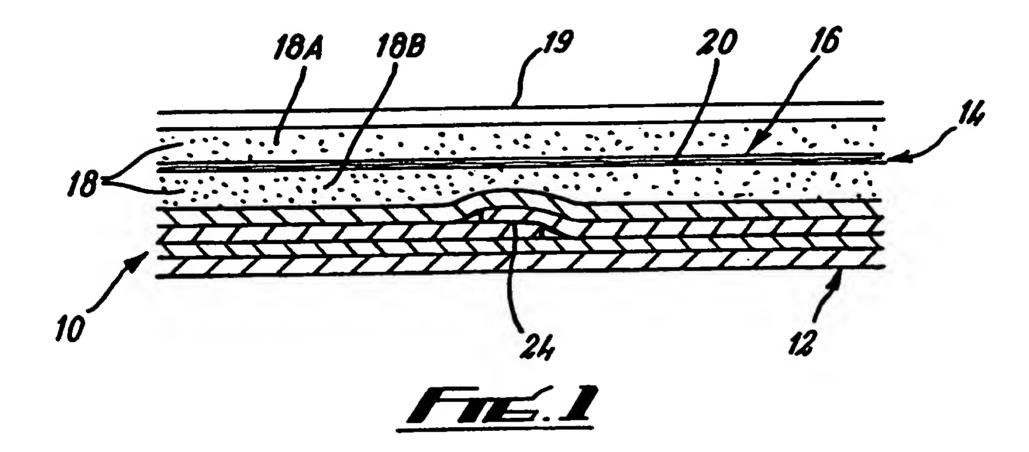
- 1. A pressure transmitter comprising an elastomeric material which is partially or at least partially cured at a temperature not exceeding 100°C.
- 2. A pressure transmitter for use in the production of composite components, the pressure transmitter comprising an inner region of a reinforcement and an outer region of an elastomeric material, the elastomeric material being partially or, at least partially, curable at a temperature not exceeding 100°C.
- 3. A pressure transmitter as claimed in claim 1 or claim 2, wherein the elastomeric material is partially or, at least partially, curable at 20 70°C.
- A pressure transmitter as claimed in claim 3 wherein the elastomeric material is partially or, at least partially, curable at 65°C.
- 5. A pressure transmitter as claimed in claim 3 or claim 4, wherein the elastomeric material is further cured to reach a state of being substantially fully cured at a temperature of at least 170°C.
- 6. A pressure transmitter as claimed in claim 5, wherein the elastomeric material is further cured to reach a state of being substantially fully cured at a temperature of 175°C.
- 7. A pressure transmitter as claimed in claim 3 or claim 4, wherein the elastomeric material is partially cured during a period of 14 hours.
- 8. A pressure transmitter as claimed in claim 5 or claim 6, wherein the elastomeric material is further cured for a period of at least 2 hours.
- 9. A pressure transmitter as claimed in any one of the preceding claims, wherein the elastomeric material is a silicone elastomer, an acrylate or a

#### fluoroelastomer.

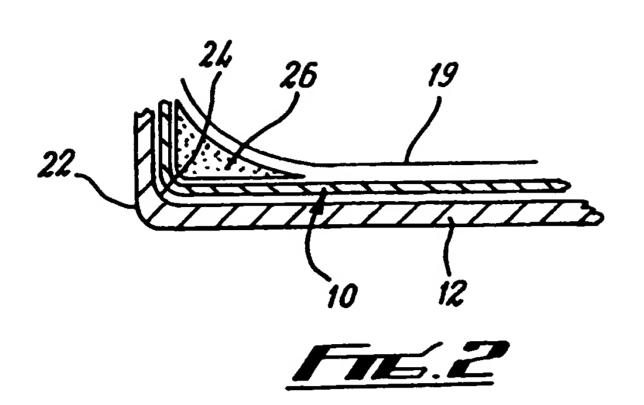
- A pressure transmitter as claimed in any of claims 2 to 9, wherein the 10. resin material is curable at a temperature not exceeding 100°C.
- A pressure transmitter as claimed in any one of claims 2 to 10, wherein 11. the curable prepreg reinforcement in the form of resin pre-impregnated fibres defining a plurality of interstitial spaces, wherein the elastomeric material of the outer region substantially fills the interstitial spaces.
- A pressure transmitter for use in the production of composite 12. components, the pressure transmitter comprising an inner region of a curable reinforcement and an outer region of elastomeric material, the curable reinforcement comprising resin pre-impregnated onto fibres, said resin being partially, or at least partially, curable at a temperature not exceeding 100°C.
- A pressure transmitter as claimed in claim 12, wherein the elastomeric 13. material cures or at least partially cures at a temperature greater than the cure or partial cure temperature of the curable reinforcement.
- A pressure transmitter as claimed in claim 13, wherein the elastomeric 14. material is partially, or at least partially, cured at a temperature in the range of 20 - 70°C.
- A pressure transmitter as claimed in claim 14, wherein the elastomeric 15. material is partially, or at least partially, cured at a temperature of 65°C.
- A pressure transmitter as claimed in any of claims 13 to 15, wherein the 16. elastomeric material is partially cured during a period of 14 hours.
- A pressure transmitter as claimed in any of claims 13 to 16, wherein the 17. elastomeric material is further cured to reach a state of full cure at a temperature of at least 170°C.

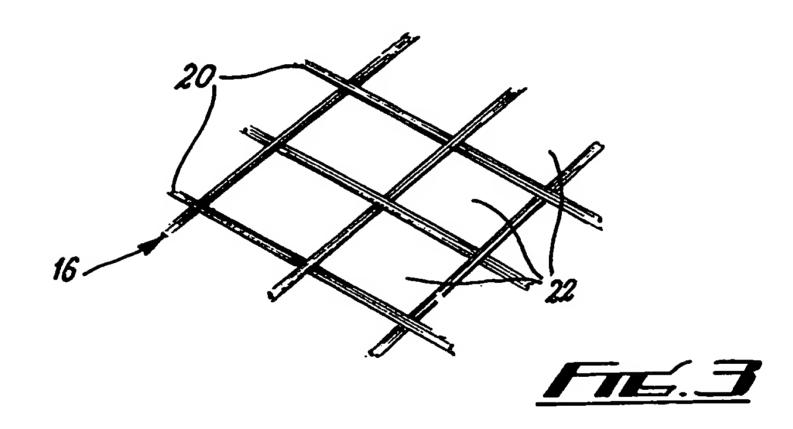
- A pressure transmitter as claimed in claim 17, wherein the elastomeric 18. material is further cured to reach a state of full cure at a temperature of 175 C.
- A pressure transmitter as claimed in claim 17 or claim 18, wherein the 19. elastomeric material is further cured during a period of 2 hours.
- A pressure transmitter for use in the production of composite 20. components, the pressure transmitter comprising an inner region of a curable prepreg reinforcement and an outer region of an elastomeric material, the curable prepreg reinforcement being in the form of resin pre-impregnated fibres defining a plurality of interstitial spaces, where in the elastomeric material of the outer region substantially fills the interstitial spaces.
- A method of curing a pressure transmitter comprising providing a 21. pressure transmitter as claimed in any one of the preceding claims, and thereafter at least partially curing the pressure transmitter at a temperature not exceeding 100°C while supported on a mould which need not be capable of withstanding temperatures of greater than 100°C.
- A pressure transmitter substantially as hereinbefore described with 22. reference to Fig. 1 or Fig. 2 or Figs. 3 and 4 of the accompanying drawings.
- A method of curing a pressure transmitter substantially as hereinbefore 23. described.
- Any novel subject matter or combination including novel subject matter 24. disclosed herein, whether or not within the scope of or relating to the same invention as any of the preceding claims.

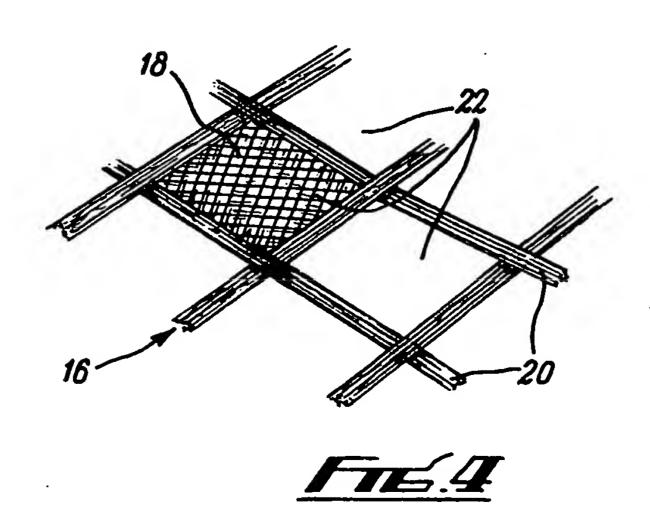
1/2











#### INTERNATIONAL SEARCH REPORT

this tional Application No PCT/GB 98/01270

A. CLASSIFICATION OF SUBJECT MATTER
IPC 6 B29C43/36 B29C33/40 //B29K21:00,B29K105:10 B30B15/06 According to International Patent Classification (IPC) or to both national classification and IPC **B. FIELDS SEARCHED** Minimum documentation searched (classification system followed by classification symbols) B29C B29D B30B Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched Electronic data base consulted during the international search (name of data base and, where practical, search terms used) C. DOCUMENTS CONSIDERED TO BE RELEVANT Relevant to claim No. Citation of document, with indication, where appropriate, of the relevant passages US 4 770 928 A (GAWOROWSKI ANDREW J ET 1-24 X AL) 13 September 1988 see the whole document 20,22-24 EP 0 318 415 A (UNITED TECHNOLOGIES CORP) X 31 May 1989 see the whole document 1,2,5,6, A 8,9, 11-13, 17-19,21 EP 0 491 646 A (UNITED TECHNOLOGIES CORP) 20,22-24 24 June 1992 1,2,5,6, see column 6, line 37 - column 9, line 9; 8,9, claims; figures 11-13, 17-19,21 Patent family members are listed in annex. Further documents are listed in the continuation of box C. Special categories of cited documents : "I" later document published after the international filling date or priority date and not in conflict with the application but "A" document defining the general state of the art which is not cited to understand the principle or theory underlying the considered to be of particular relevance invention "E" earlier document but published on or after the international "X" document of particular relevance; the claimed invention filling date cannot be considered novel or cannot be considered to involve an inventive step when the document is taken alone "L" document which may throw doubts on priority claim(s) or which is cited to establish the publication date of another "Y" document of particular relevance; the claimed invention citation or other special reason (as specified) cannot be considered to involve an inventive step when the document is combined with one or more other such docu-"O" document referring to an oral disclosure, use, exhibition or ments, such combination being obvious to a person skilled other means in the art. "P" document published prior to the international filing date but "&" document member of the same patent family later than the priority date claimed Date of mailing of the international search report Date of the actual completion of theinternational search 26/08/1998 19 August 1998 Authorized officer Name and mailing address of the ISA European Patent Office, P.B. 5818 Patentiaan 2 NL - 2280 HV Rijswijk Tel. (+31-70) 340-2040, Tx. 31 651 epo nl, Labeeuw, R Fax: (+31-70) 340-3016

# INTERNATIONAL SEARCH REPORT

Inti Ional Application No PCT/GB 98/01270

Column	0.70		1 C1/ GD 30/ U12/ U
US 4 824 631 A (YEAGER JAMES F) 25 April 1,5,6,8, 1989 see the whole document  US 3 990 930 A (SCHMIT GEORGES) 9 November 1,2,5,6, 11,12, 17,18, 20-24  see the whole document  US 5 324 376 A (WEIMER RUSSELL ET AL) 28 June 1994  see column 2, line 49 - column 4, line 39; claims 7-14; figures  EP 0 491 651 A (UNITED TECHNOLOGIES CORP) 24 June 1992  see the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 see column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  1,2,12, 20,22-24		······································	Relevant to claim No.
1989   9,22-24			
11,12, 17,18, 20-24  see the whole document  US 5 324 376 A (WEIMER RUSSELL ET AL) 28 June 1994  see column 2, line 49 - column 4, line 39; claims 7-14; figures  EP 0 491 651 A (UNITED TECHNOLOGIES CORP) 24 June 1992  see the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 see column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  11,12, 17,18, 20-24  1,2, 11-13, 20,22-24  1,2,2, 11-13, 20-24  1,2,2-24  1,2,12, 20,22-24	A	1989	
See the whole document  US 5 324 376 A (WEIMER RUSSELL ET AL) 28 June 1994  See column 2, line 49 - column 4, line 39; claims 7-14; figures  EP 0 491 651 A (UNITED TECHNOLOGIES CORP) 24 June 1992  See the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 See column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  1,2,12, 20,22-24	4		11,12, 17,18,
June 1994  see column 2, line 49 - column 4, line 39; claims 7-14; figures  EP 0 491 651 A (UNITED TECHNOLOGIES CORP) 24 June 1992  see the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 see column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  11-13, 20,22-24  1,2,9, 11-13, 20-24  1,2,9, 11-13, 20-24  1,2,12, 20-24	1	see the whole document	20 21
see column 2, line 49 - column 4, line 39; claims 7-14; figures  EP 0 491 651 A (UNITED TECHNOLOGIES CORP) 24 June 1992  see the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 see column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  LTD) 10 January 1996  LTD) 10 January 1996	A		11-13,
24 June 1992  see the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 see column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  11-13, 20-24  1,22-24  1,22-24  1,2,12, 20,22-24			
See the whole document  EP 0 572 248 A (BRITISH AEROSPACE) 1 December 1993 See column 4, line 19 - column 5, line 19; claims; figures  A GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  1,2,12, 20,22-24	A		11-13,
December 1993 see column 4, line 19 - column 5, line 19; claims; figures  GB 2 290 742 A (ADVANCED COMPOSITES GROUP LTD) 10 January 1996  1,2,12, 20,22-24		see the whole document	
LTD) 10 January 1996 20,22-24	A	December 1993 see column 4, line 19 - column 5, line 19;	1,22-24
	A	LTD) 10 January 1996	
		•	
		•	
. 1			

# INTERNATIONAL SEARCH REPORT

Information on patent family members

trite onal Application No PCT/GB 98/01270

Patent document cited in search repo		Publication date	Patent family member(s)		Publication date
US 4770928	Α	13-09-1988	EP	0342286 A	23-11-1989
			DE	3888679 D	28-04-1994
			DE	3888679 T	03-11-1994
EP 0318415	Α	31-05-1989	JP	1069637 A	15-03-1989
			US	5071338 A	10-12-1991
			US	5217669 A	08-06-1993
EP 0491646	A	24-06-1992	US	5152949 A	06-10-1992
			CA	2054066 A	20-06-1992
			DE	69109255 D	01-06-1995
			DE	69109255 T	24-08-1995
			JP	4294113 A	19-10-1992
US 4824631	Α	25-04-1989	NONE		
US 3990930	Α	09-11-1976	NONE	•	
US 5324376	Α	28-06-1994	US	5250142 A	05-10-1993
EP 0491651	Α	24-06-1992	US	5126092 A	30-06-1992
			CA	2056330 A	20-06-1992
			DE	69111775 D	07-09-1995
	•		DE	69111775 T	11-01-1996
			JP	4301410 A	26-10-1992
		· .	US	5286438 A	15-02-1994
EP 0572248	Α	01-12-1993	NONE		
GB 2290742	Α	10-01-1996	AU	2748995 A	25-01-1996
			. WO	9600642 A	11-01-1996